Numerical investigations of the stepped double torsion fracture toughness specimen

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This study aims at investigating the fracture behaviour of double torsion specimens using the finite element method. Typical double torsion tests encompass a series of constant-thickness specimens to evaluate the material plane strain fracture toughness. In contrast, the concept of using a novel variable thickness stepped specimen aims at deducing the fracture toughness using a single specimen. In this work, the feasibility of this approach is examined and the effect of the number of steps and fracture thickness in a specimen upon the resulting conditional stress intensity factor is evaluated. The finite element models employed experimentally determined values of the fracture load to evaluate the conditional stress intensity factor of the specimen. Finite element predictions were compared with earlier experimental results using both cast aluminium silicon alloy and gray cast iron specimens and good matching was observed between experimental results and numerical predictions.

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1. Introduction

The double torsion test (DT) has been introduced in the late sixties as a relatively simple and inexpensive technique to evaluate the fracture mechanics characteristics of both metallic and nonmetallic materials. Specifically, DT testing has been extensively used to study both plane strain fracture toughness \( K_{IC} \) evaluation and slow crack growth characteristics in situations where standard size requirements cannot be satisfied, such as thin sheets. The pioneering work of Outwater and Gerry [1,2] and Kies and Clark [3] laid the experimental foundation for this test methodology. The main characteristic of this technique is the independence of the stress intensity factor \( K_I \) from the crack length \( a \), thus making this test ideal for situations where it is difficult to measure the crack length during testing as in opaque materials or elevated temperature testing. Extensive research has since been devoted to investigate the various aspects of this technique [4–6].

Fig. 1 depicts a typical test configuration, in which a rectangular specimen with length \( L \), width \( W \), and thickness \( t \) rests on roller supports and is loaded by a force \( P \) in a four-point bending setup. This loading configuration enables a crack of length \( a \) along the symmetry plane of the two halves to incrementally propagate by increments \( \Delta a \). Hence, the developed two branches are loaded in torsion along the crack length by a torsional moment \( M \). Analytical treatment of this problem assumes that the width \( W \) is much larger than the thickness \( t \) and that the uncracked region is rigid, especially for moderately long cracks. These assumptions enable the well-known analytical solution of non-circular bars torsion to be applied to this problem [7–10]. Hence, the conditional stress intensity factor is given by:
\[ K_Q = \begin{cases} \frac{P_c W_m}{2} \sqrt{\frac{3(1-t)}{t^2 W^2}} & \text{for plane stress} \\ \frac{P_c W_m}{2} \sqrt{\frac{3}{t^2 W^2(1-t)}} & \text{for plane strain} \end{cases} \]  

where \( \psi \) is a thickness correction function for thick beams [10] given by:

\[ \psi = 1 - 0.6302 \left( \frac{2t_c}{W} \right) + 1.2 \left( \frac{2t_c}{W} \right) e^{-\left( \frac{2t_c}{W} \right)} \]  

Proper understanding of the mode of fracture and the applicable analytical model (plane stress/strain) is pivotal to standardizing the DT test. While it seems that the crack propagation is mode-III dominated, close agreement was found between experiments and analytical results using mode-I analysis [7,8]. Furthermore, previous studies employed both plane stress and plane strain models to evaluate the critical stress intensity factors [8,11] and compare to fracture toughness estimates using standard tests. A closer examination of Eq. (1) shows that the difference between the two models is dependent upon Poisson’s ratio of the material. The validity of these assumptions will be assessed in the present work using three-dimensional finite element analyses.

Trantina [12] conducted the earliest three-dimensional FE simulations of the DT specimen using a bilinear crack front to approximate the actual curved front. This study aimed to investigate the effect of specimen geometry and crack front shape upon the resulting \( K_I \) values. In his work, he employed singular crack elements with \( 1/\sqrt{r} \) strain singularity around the crack tip and evaluated the stress intensity factor using the displacement matching method. In order to alleviate the steep stress gradient near the crack tip and due to the limited computational resources at that time, a small radius fillet was introduced at the crack tip. It was found that choosing specimen dimensions such that \( a/W > 0.55 \) and \( (L-a)/W > 0.65 \) results in a very close agreement between numerical and analytical \( K_I \) predictions (within 5%) in the lower half of the specimen thickness. These results were later confirmed by White [13] using line spring approximate finite element analysis and the same bilinear crack profile to evaluate \( K_I \) for the DT specimen. The work of White was further extended to examine the more realistic quarter-ellipse crack profiles and it was found that the analytical estimate in Eq. (1) overestimates \( K_I \) values by 10–20%. Tseng and Berry [14] established the following bounds on the validity of the DT test: \( L = 2 \ W \) and \( 1/12 \leq t/W \leq 1/6 \). They also
investigated the effect of top and bottom grooves and found the presence of grooves to have marginal effects on the resulting $K_I$ levels.

More recently, Ciccotti and his co-workers [15,16] performed three-dimensional finite element analyses of the DT test using crack advance technique. This technique is particularly accurate and does not impose excessive mesh refinement requirements near the crack tip, yet it is rather demanding in terms of the number of simulations with different crack lengths to evaluate the change in the strain energy release rate (SERR). They studied the influence on the SERR of the specimen width and length, the presence and shape of the bottom groove, the presence of an initial notch, and estimated the curved shape of the fracture front.

The concept of stepped DT specimens aims at deducing the critical stress intensity factor using a single specimen with varying $t_c/t$ ratio, rather than several specimens of different thicknesses. This approach has been implemented experimentally for brittle materials using gray cast iron [17]. The objective of the present study is to investigate the effect of the fracture thickness ($t_c$) and the number of steps of the stepped DT specimen upon the resulting conditional stress intensity factor using three-dimensional finite element analysis. In the present work, stress intensity factor is evaluated using virtual crack extension approach [18], in which the SERR is evaluated by virtually relocating a set of elements in the immediate vicinity of the crack by a distance of $\Delta a$. The main advantage of this technique is that it obviates the need for successive mesh generation to change the crack length to calculate the SERR.

2. Finite element modeling

2.1. Model geometry

The dimensions of both constant-thickness and stepped DT specimens were chosen similar to those in [17,19] to enable comparing the numerical and experimental results. For constant-thickness specimens, the width between the two rollers $W_m = 62$ mm, length $L = 150$ mm, and total width $W = 75$ mm. The specimen is grooved along its longitudinal midplane on both its top and bottom faces. The reduced thickness $t_c$ was varied in order to investigate the effect of the fracture thickness ratio ($t_c/t$) upon the resulting conditional stress intensity factor $K_Q$. Table 1 shows the various levels of ($t_c/t$) under consideration. In total, eight specimens are considered with $0.267 < t_c/t < 0.933$.

Two groups of three- and five-steps DT specimen configurations are considered herein for FEA with their respective geometries and dimensions as depicted in Fig. 2 and Tables 2 and 3. The specimen width between the two rollers is $W_m = 100$ mm. The minimum thickness $t_c$ in Fig. 2 is varied in intervals along the specimen length such that it is experimentally viable to obtain a series of values of the critical fracture load $P_c$ in Eq. (1) along different length zones in a single specimen, thereby eliminating the need for a multiple of specimens and experiments. The length of each zone $L_i$ was selected to be sufficient to prevent interaction between different zones.

Special attention was devoted to the geometry of the crack front. The actual curved crack front was approximated using a four-segment polyline over the minimum specimen thickness $t_c$ such that the inclination angle increases from $10^\circ$ to $30^\circ$ over four equal vertical distances $t_c/4$, as shown in Fig. 3. This approximation has been successfully implemented earlier in [15] and is adopted to overcome three-dimensional mesh generation issues of curved crack fronts.

2.2. Mesh generation

Mesh generation is the most intricate part in the finite element analysis of a crack geometry. The most common technique for crack meshing is to implement a spider-web mesh around the crack front, with quarter-point degenerate quadratic elements on the first row of elements surrounding the crack front. These are modified 20-node three-dimensional solid elements, where the mid-side nodes are shifted towards to the quarter-point near to the crack tip and the element is collapsed into a wedge shape along the crack front. This modification results in a 1/$\sqrt{r}$ strain singularity in the element, thereby accurately simulating the physical stress field singularity around the crack tip [12]. Fig. 4 illustrates the mesh design for a five-step DT model with a spider mesh around the crack tip. Away from the crack tip, standard 20-node hexahedral

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**Table 1**

Dimensions of constant-thickness DT cast aluminium specimens.

<table>
<thead>
<tr>
<th>Specimen</th>
<th>$W_m$ (mm)</th>
<th>$t$ (mm)</th>
<th>$t_c/t$</th>
</tr>
</thead>
<tbody>
<tr>
<td>ST1-1</td>
<td>62</td>
<td>15</td>
<td>0.27</td>
</tr>
<tr>
<td>ST1-2</td>
<td>0.4</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-3</td>
<td>0.53</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-4</td>
<td>0.67</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-5</td>
<td>0.8</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-6</td>
<td>0.87</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-7</td>
<td>0.9</td>
<td></td>
<td></td>
</tr>
<tr>
<td>ST1-8</td>
<td>0.93</td>
<td></td>
<td></td>
</tr>
</tbody>
</table>
elements with three translational degrees of freedom are used. In the present work, mesh generation has been performed using general purpose finite element analysis package [20].

Table 2
Dimensions of three-steps DT grey cast iron specimens.

<table>
<thead>
<tr>
<th>Specimen</th>
<th>W_m (mm)</th>
<th>t (mm)</th>
<th>t_c/t</th>
</tr>
</thead>
<tbody>
<tr>
<td>ST3-1</td>
<td>100</td>
<td>13</td>
<td>0.35</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.57</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.77</td>
</tr>
<tr>
<td>ST3-2</td>
<td>100</td>
<td>13</td>
<td>0.69</td>
</tr>
<tr>
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<td></td>
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<td></td>
<td></td>
<td></td>
<td>0.82</td>
</tr>
<tr>
<td>ST3-3</td>
<td>100</td>
<td>13</td>
<td>0.85</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.88</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.92</td>
</tr>
</tbody>
</table>

Table 3
Dimensions of five-steps DT grey cast iron specimens.

<table>
<thead>
<tr>
<th>Specimen</th>
<th>W_m (mm)</th>
<th>t (mm)</th>
<th>t_c/t</th>
</tr>
</thead>
<tbody>
<tr>
<td>ST5-1</td>
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<td>13.5</td>
<td>0.33</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
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<td>ST5-2</td>
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<td>0.56</td>
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<td></td>
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<td>0.71</td>
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<td></td>
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<td></td>
<td>0.75</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.91</td>
</tr>
<tr>
<td>ST5-3</td>
<td>100</td>
<td>13.6</td>
<td>0.34</td>
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<tr>
<td></td>
<td></td>
<td></td>
<td>0.59</td>
</tr>
<tr>
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<td></td>
<td></td>
<td>0.73</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.79</td>
</tr>
<tr>
<td></td>
<td></td>
<td></td>
<td>0.92</td>
</tr>
</tbody>
</table>

Fig. 2. Schematic of stepped double torsion models: (a) five-step and (b) three-step.

Fig. 3. Polylne discretization of the curved crack front.
2.3. Boundary conditions

The DT specimen geometry and loading conditions depicted in Fig. 1 makes it possible to include only one-half of the specimen in the FE model, as shown in Fig. 5 for the constant-thickness model. Similar conditions can be applied to the stepped model. In this figure, two equal and opposite vertical forces of magnitude $P_c/2$ were applied at the locations of the loading fixture and roller support, respectively, to simulate load control conditions. Symmetry boundary conditions were applied along the uncracked ligament of the specimen such that the nodes on this ligament can only move in the $x$–$y$ plane. It was sufficient to fully constrain one node on the far end of the model in all directions to prevent rigid body motion; this node was chosen at the midplane of the model as shown in Fig. 5.

2.4. Material properties

The material properties for the finite element simulation were chosen to match those of the experimental work in [17] and [19]. Cast aluminium silicon alloy and gray cast iron were specifically chosen for their low ductility and proximity to brittle fracture behaviour. Table 4 summarizes the mechanical properties of these two alloys. It shows that both alloys exhibit almost negligible ductility and their fracture behaviour is hence expected to be accurately described within the framework of linear elastic fracture mechanics. Furthermore, both materials were assumed to follow a linear elastic isotropic...
behaviour. Experimental evaluation of the plane strain fracture toughness was made using ASTM E399-90\[21\] standard three point bend test method. Fig. 6a and b depicts the experimental setup used in[17]. Proprietary load cell and displacement transducer were fabricated and calibrated to ensure high measurement accuracy of the load–displacement pairs. The base plate was mounted on a centering pin on the testing machine to ensure co-axial loading and displacement control test protocol was applied at a constant rate of 1.5\[15\]/C2\[15\]10\[15\]/C0\[15\]3 mm. The test specimen was carefully positioned on the supporting rollers to ensure symmetrical loading around the center of the test specimen.

3. Results and discussion

3.1. Uniform-thickness DT model

The conditional stress intensity factor (\(K_Q\)) was evaluated analytically and numerically using eight constant-thickness specimens made of cast aluminium. The thickness ratio for these specimens (\(t_c/t\)) was gradually increased from 0.27 to 0.93. The analytical plane stress and plane strain estimates are obtained using Eq. (1). The values of \(K_Q\) are normalized with respect to standard \(K_{IC}\) value obtained using ASTM compact tension specimen (see Table 4). The experimental value of the critical load \(P_c\) was used to evaluate \(K_Q\) in both the analytical and finite element computations. Table 5 illustrates the variation of \(P_c\) with increased \(t_c/t\) ratio.

Fig. 7 depicts the variation of normalized \(K_Q\) with the fracture thickness ratio (\(t_c/t\)) using the calibration equation (plane stress and plane strain) and finite element analysis. In this figure, it is observed that both plane stress and plane strain estimates are very close to each other. In fact, plane strain estimates are higher by a factor of \(\sqrt{1-v^2}\), which amounts to 4.83% in the current problem. It is also observed that all approaches exhibit a similar pattern, in which the values of \(K_Q\) gradually increase with (\(t_c/t\)) until they reach a peak value at \(t_c/t \approx 0.7\). Then, \(K_Q\) asymptotically drops to approach the standard \(K_{IC}\) value. This pattern can be explained by considering the transition from plane stress to plane strain behaviour. For small \(t_c/t\) values, the fracture behaviour is strictly plane stress, and increasing the thickness is associated with a comparable increase in critical load \(P_c\) levels. As the specimen critical thickness increases, marked stress triaxiality is observed due to increased constrained material volume, thereby leading to plane strain conditions and associated reduction in \(K_Q\) levels.

<table>
<thead>
<tr>
<th>Model #</th>
<th>(t_c/t)</th>
<th>(P_c) (kN)</th>
<th>(K_Q) (Exp.) (MPa √m)</th>
<th>(K_Q) (FEA) (MPa √m)</th>
</tr>
</thead>
<tbody>
<tr>
<td>ST1-1</td>
<td>0.27</td>
<td>5.2</td>
<td>10.33</td>
<td>8.56</td>
</tr>
<tr>
<td>ST1-2</td>
<td>0.4</td>
<td>8.9</td>
<td>14.5</td>
<td>11.01</td>
</tr>
<tr>
<td>ST1-3</td>
<td>0.53</td>
<td>11.1</td>
<td>15.63</td>
<td>12.22</td>
</tr>
<tr>
<td>ST1-4</td>
<td>0.67</td>
<td>16.1</td>
<td>20.25</td>
<td>16.86</td>
</tr>
<tr>
<td>ST1-5</td>
<td>0.8</td>
<td>16.3</td>
<td>18.74</td>
<td>16.27</td>
</tr>
<tr>
<td>ST1-6</td>
<td>0.87</td>
<td>12.8</td>
<td>14.08</td>
<td>13.04</td>
</tr>
<tr>
<td>ST1-7</td>
<td>0.9</td>
<td>11.5</td>
<td>12.48</td>
<td>11.24</td>
</tr>
<tr>
<td>ST1-8</td>
<td>0.93</td>
<td>12</td>
<td>12.73</td>
<td>11.48</td>
</tr>
</tbody>
</table>
Comparing the plane strain fracture toughness estimates \( K_{IC} = 12.1 \text{ MPa} \sqrt{m} \) in [19] with \( K_Q \) obtained using FEA (11.24 MPa \( \sqrt{m} \)), a good agreement is observed between both approaches. It is noteworthy to observe that LEFM-based FEA estimates are consistently lower than the experimental values. Aside of common errors in experimental measurements, the most plausible source of discrepancy between these values is the development of a plastic zone around the crack tip, which leads to limited crack blunting. In this case, using \( J \)-integral, coupled with an elasto-plastic material model, should give a better estimate of the material fracture toughness. Meanwhile, the analytical estimates, both plane stress and plane strain, overestimates \( K_Q \) value compared to both experiments results and FE predictions due to the numerous assumptions inherent to this equation. For instance, assuming that the uncracked ligaments of the specimen are rigid artificially stiffens the specimen and yields higher critical fracture load value \( P_c \). This has already been pointed out in an earlier study [15].

### 3.2. Variable thickness stepped DT model

The conditional stress intensity factor \( (K_Q) \) values were evaluated using three- and five-steps DT finite element models having fracture thickness ratios \( t_c/t \) that range between 0.35 and 0.92. The numerical predictions were compared with the experimental values in [17]. Fig. 8 depicts the variation of \( K_Q \) with \( t_c/t \) for three-step DT specimens for all ratios considered in Table 2. In this figure, \( K_Q \) is normalized with respect to standard \( K_{IC} \) value obtained in [17]. Three separate lines indicate three different models, each with a different set of gradually increasing \( t_c/t \). This was necessary to enable comparison of

![Normalized conditional SIF (K_Q) predictions using various techniques for uniform-thickness DT specimen.](image)

**Fig. 7.** Normalized conditional SIF \( (K_Q) \) predictions using various techniques for uniform-thickness DT specimen.

![Normalized conditional SIF (K_Q) predictions using three-step DT specimen.](image)

**Fig. 8.** Normalized conditional SIF \( (K_Q) \) predictions using three-step DT specimen.

### Table 6

<table>
<thead>
<tr>
<th>Specimen</th>
<th>( t_c/t )</th>
<th>( P_c ) (kN)</th>
<th>( K_Q ) (Exp.) (MPa ( \sqrt{m} ))</th>
<th>( K_Q ) (FEA) (MPa ( \sqrt{m} ))</th>
</tr>
</thead>
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<tr>
<td>ST3-1</td>
<td>0.35</td>
<td>8.9</td>
<td>25.72</td>
<td>23.55</td>
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<tr>
<td></td>
<td>0.57</td>
<td>13</td>
<td>29.57</td>
<td>27.78</td>
</tr>
<tr>
<td></td>
<td>0.77</td>
<td>21.5</td>
<td>41.96</td>
<td>41.53</td>
</tr>
<tr>
<td>ST3-2</td>
<td>0.69</td>
<td>16</td>
<td>34.46</td>
<td>30.54</td>
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<tr>
<td></td>
<td>0.75</td>
<td>20.4</td>
<td>42.11</td>
<td>38.94</td>
</tr>
<tr>
<td></td>
<td>0.82</td>
<td>21</td>
<td>41.68</td>
<td>39.67</td>
</tr>
<tr>
<td>ST3-3</td>
<td>0.85</td>
<td>16.8</td>
<td>31.88</td>
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<tr>
<td></td>
<td>0.88</td>
<td>17.3</td>
<td>32.11</td>
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<td></td>
<td>0.92</td>
<td>17.8</td>
<td>32.34</td>
<td>32.03</td>
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</table>
experimental and finite element analysis. Similar to the plain double torsion specimens, $K_Q$ value gradually increased with $\left(\frac{t_c}{t}\right)$ until it reached its peak value at $\frac{t_c}{t} = 0.769$. Then, $K_Q$ value asymptotically decreased approaching $K_{IC}$ value as $\frac{t_c}{t} = 0.846$. Comparing plane strain fracture toughness estimate (31.29 MPa $\sqrt{m}$) obtained [17] with $K_Q$ value using FEA (30.14 MPa $\sqrt{m}$), a good agreement is observed. Furthermore, FEA predictions were in good agreement with the experimental value of the critical stress intensity factor at fracture throughout the thickness range $\frac{t_c}{t}$ under consideration. Table 6 presents a comparison between the $K_Q$ obtained by experimental work in [17] and using FEA.

Fig. 9 depicts the variation of normalized $K_Q$ with $\frac{t_c}{t}$ for five-step DT specimens using both finite element analysis and the experimental values in [17]. The figure presents these results for the three specimens under consideration, shown in Table 3. In contrast to the three-step DT specimens, it was possible here to cover the full range of $\frac{t_c}{t}$ values under investigation (0.33–0.92) using one specimen since the change in $\frac{t_c}{t}$ value over each zone was less abrupt compared to the three-step model. The results show a consistent pattern of $K_Q$ and $\frac{t_c}{t}$ relation, which is similar to that observed in constant-thickness and three-step specimens (see Table 7).

### 4. Conclusions

In this study, three-dimensional finite element analysis was conducted of constant and variable thickness stepped double torsion specimens. The objective was to obtain the material critical stress intensity factor and to compare it with the plane strain fracture toughness estimates using standard tests. Comparison of numerical and experimental results showed a good agreement between the two approaches for single, three-, and five-steps double torsion specimens. The variation of the conditional stress intensity factor with thickness showed a distinctive pattern of monotonic increase until a thickness ratio of $\approx 0.7$ is achieved, after which its value asymptotically approaches the experimental plane strain fracture toughness value. This pattern has been explained in view of mixed plane stress/strain fracture behaviour. Slight deviation existed between numerical and experimental results, which are attributed to the discretised representation of the curved crack front and the development of limited plastic zone ahead of the crack tip, which was not accounted for in the finite element simulations. The results also showed the stepped double torsion test specimen is promising in determining the fracture toughness of the materials, both analytically and experimentally.
References